16.323 Principles of Optimal Control Spring 2008

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#### 16.323 Lecture 16

Model Predictive Control

- Allgower, F., and A. Zheng, Nonlinear Model Predictive Control, Springer-Verlag, 2000.
- Camacho, E., and C. Bordons, Model Predictive Control, Springer-Verlag, 1999.
- Kouvaritakis, B., and M. Cannon, Non-Linear Predictive Control: Theory & Practice, IEE Publishing, 2001.
- Maciejowski, J., Predictive Control with Constraints, Pearson Education POD, 2002.
- Rossiter, J. A., Model-Based Predictive Control: A Practical Approach, CRC Press, 2003.

# MPC

- Planning in Lecture 8 was effectively "open-loop"
  - Designed the control input sequence  $\mathbf{u}(t)$  using an assumed model and set of constraints.
  - Issue is that with modeling error and/or disturbances, these inputs will not necessarily generate the desired system response.
- Need a "closed-loop" strategy to compensate for these errors.
  - Approach called **Model Predictive Control**
  - Also known as receding horizon control
- Basic strategy:
  - At time k, use knowledge of the system model to design an input sequence

$$\mathbf{u}(k|k), \mathbf{u}(k+1|k), \ \mathbf{u}(k+2|k), \ \mathbf{u}(k+3|k), \dots, \mathbf{u}(k+N|k)$$

- over a finite horizon N from the current state  $\mathbf{x}(k)$
- Implement a fraction of that input sequence, usually just first step.
- Repeat for time k + 1 at state  $\mathbf{x}(k + 1)$



Figure by MIT OpenCourseWare.

- Note that the control algorithm is based on numerically solving an optimization problem at each step
  - Typically a constrained optimization
- Main advantage of MPC:
  - Explicitly accounts for system constraints.
  - $\diamondsuit$  Doesn't just design a controller to keep the system away from them.
  - Can easily handle nonlinear and time-varying plant dynamics, since the controller is explicitly a function of the model that can be modified in real-time (and plan time)
- Many commercial applications that date back to the early 1970's, see http://www.che.utexas.edu/~qin/cpcv/cpcv14.html
  - Much of this work was in process control very nonlinear dynamics, but not particularly fast.
- As computer speed has increased, there has been renewed interest in applying this approach to applications with faster time-scale: trajectory design for aerospace systems.



Figure by MIT OpenCourseWare.

# **Basic Formulation**

• Given a set of plant dynamics (assume linear for now)

$$\begin{aligned} \mathbf{x}(k+1) &= A\mathbf{x}(k) + B\mathbf{u}(k) \\ \mathbf{z}(k) &= C\mathbf{x}(k) \end{aligned}$$

and a cost function

$$J = \sum_{j=0}^{N} \{ \|\mathbf{z}(k+j|k)\|_{R_{zz}} + \|\mathbf{u}(k+j|k)\|_{R_{uu}} \} + F(\mathbf{x}(k+N|k))$$

 $- \|\mathbf{z}(k+j|k)\|_{R_{\mathrm{xx}}}$  is just a short hand for a weighted norm of the state, and to be consistent with earlier work, would take

$$\|\mathbf{z}(k+j|k)\|_{R_{zz}} = \mathbf{z}(k+j|k)^T R_{zz} \mathbf{z}(k+j|k)$$

- $\ F(\mathbf{x}(k+N|k))$  is a terminal cost function
- Note that if N → ∞, and there are no additional constraints on z or u, then this is just the discrete LQR problem solved on page 3–14.
  - Note that the original LQR result could have been written as just an input control sequence (feedforward), but we choose to write it as a linear state feedback.
  - In the nominal case, there is no difference between these two implementation approaches (feedforward and feedback)
  - But with modeling errors and disturbances, the state feedback form is much less sensitive.

 $\Rightarrow$  This is the main reason for using feedback.

 Issue: When limits on x and u are added, we can no longer find the general solution in analytic form ⇒ must solve it numerically.

- However, solving for a very long input sequence:
  - Does not make sense if one expects that the model is wrong and/or there are disturbances, because it is unlikely that the end of the plan will be implemented (a new one will be made by then)
  - Longer plans have more degrees of freedom and take much longer to compute.
- Typically design using a small  $N \Rightarrow$  short plan that does not necessarily achieve all of the goals.
  - Classical hard question is how large should N be?
  - If plan doesn't reach the goal, then must develop an estimate of the remaining cost-to-go
- Typical problem statement: for finite N (F = 0)

$$\min_{u} J = \sum_{j=0}^{N} \{ \|\mathbf{z}(k+j|k)\|_{R_{zz}} + \|\mathbf{u}(k+j|k)\|_{R_{uu}} \}$$
  
s.t.  $\mathbf{x}(k+j+1|k) = A\mathbf{x}(k+j|k) + B\mathbf{u}(k+j|k)$   
 $\mathbf{x}(k|k) \equiv \mathbf{x}(k)$   
 $\mathbf{z}(k+j|k) = C\mathbf{x}(k+j|k)$ 

and  $|\mathbf{u}(k+j|k)| \leq u_m$ 

• Consider converting this into a more standard optimization problem.

$$\mathbf{z}(k|k) = C\mathbf{x}(k|k)$$
$$\mathbf{z}(k+1|k) = C\mathbf{x}(k+1|k) = C(A\mathbf{x}(k|k) + B\mathbf{u}(k|k))$$
$$= CA\mathbf{x}(k|k) + CB\mathbf{u}(k|k)$$
$$\mathbf{z}(k+2|k) = C\mathbf{x}(k+2|k)$$
$$= C(A\mathbf{x}(k+1|k) + B\mathbf{u}(k+1|k))$$
$$= CA(A\mathbf{x}(k|k) + B\mathbf{u}(k|k)) + CB\mathbf{u}(k+1|k)$$
$$= CA^{2}\mathbf{x}(k|k) + CAB\mathbf{u}(k|k) + CB\mathbf{u}(k+1|k)$$
$$\vdots$$
$$\mathbf{z}(k+N|k) = CA^{N}\mathbf{x}(k|k) + CA^{N-1}B\mathbf{u}(k|k) + \cdots$$
$$+ CB\mathbf{u}(k+(N-1)|k)$$

• Combine these equations into the following:

$$\begin{bmatrix} \mathbf{z}(k|k) \\ \mathbf{z}(k+1|k) \\ \mathbf{z}(k+2|k) \\ \vdots \\ \mathbf{z}(k+N|k) \end{bmatrix} = \begin{bmatrix} C \\ CA \\ CA^{2} \\ \vdots \\ CA^{N} \end{bmatrix} \mathbf{x}(k|k)$$

$$+ \begin{bmatrix} 0 & 0 & 0 & \cdots & 0 \\ CB & 0 & 0 & \cdots & 0 \\ CAB & CB & 0 & 0 & 0 \\ CAB & CB & 0 & 0 & 0 \\ \vdots \\ CA^{N-1}B & CA^{N-2}B & CA^{N-3}B & \cdots & CB \end{bmatrix} \begin{bmatrix} \mathbf{u}(k|k) \\ \mathbf{u}(k+1|k) \\ \vdots \\ \mathbf{u}(k+N-1|k) \end{bmatrix}$$

• Now define

$$Z(k) \equiv \begin{bmatrix} \mathbf{z}(k|k) \\ \vdots \\ \mathbf{z}(k+N|k) \end{bmatrix} \quad U(k) \equiv \begin{bmatrix} \mathbf{u}(k|k) \\ \vdots \\ \mathbf{u}(k+N-1|k) \end{bmatrix}$$

then, with  $\mathbf{x}(k|k) = \mathbf{x}(k)$ 

$$Z(k) = G\mathbf{x}(k) + HU(k)$$

• Note that

$$\sum_{j=0}^{N} \mathbf{z}(k+j|k)^{T} R_{zz} \mathbf{z}(k+j|k) = Z(k)^{T} W_{1} Z(k)$$

with an obvious definition of the weighting matrix  $W_1$ 

• Thus

$$Z(k)^{T}W_{1}Z(k) + U(k)^{T}W_{2}U(k)$$
  
=  $(G\mathbf{x}(k) + HU(k))^{T}W_{1}(G\mathbf{x}(k) + HU(k)) + U(k)^{T}W_{2}U(k)$   
=  $\mathbf{x}(k)^{T}H_{1}\mathbf{x}(k) + H_{2}^{T}U(k) + \frac{1}{2}U(k)^{T}H_{3}U(k)$ 

where

$$H_1 = G^T W_1 G, \quad H_2 = 2(\mathbf{x}(k)^T G^T W_1 H), \quad H_3 = 2(H^T W_1 H + W_2)$$

• Then the MPC problem can be written as:

$$\min_{U(k)} \tilde{J} = H_2^T U(k) + \frac{1}{2} U(k)^T H_3 U(k)$$
  
s.t.  $\begin{bmatrix} I_N \\ -I_N \end{bmatrix} U(k) \le u_m$ 

### **Toolboxes**

Key point: the MPC problem is now in the form of a standard quadratic program for which standard and efficient codes exist.
 QUADPROG Quadratic programming. %
 X=QUADPROG(H,f,A,b) attempts to solve the %
 quadratic programming problem:

min 0.5\*x'\*H\*x + f'\*x subject to: A\*x <= b
x</pre>

X=QUADPROG(H,f,A,b,Aeq,beq) solves the problem %
above while additionally satisfying the equality%
constraints Aeq\*x = beq.

- Several Matlab toolboxes exist for testing these ideas
  - MPC toolbox by Morari and Ricker extensive analysis and design tools.
  - MPCtools <sup>32</sup> enables some MPC simulation and is free www.control.lth.se/user/johan.akesson/mpctools/

 $<sup>^{32}</sup>$ Johan Akesson: "MPC<br/>tools 1.0 - Reference Manual". Technical report ISRN LUTFD2/TFRT–7613–SE, Department of Automatic Control, Lund Institute of Technology, Sweden, January 2006.

# MPC Observations 16.323 16-8

- Current form assumes that full state is available can hookup with an estimator
- Current form assumes that we can sense and apply corresponding control immediately
  - With most control systems, that is usually a reasonably safe assumption
  - Given that we must re-run the optimization, probably need to account for this computational delay - different form of the discrete model - see F&P (chapter 2)
- If the constraints are not active, then the solution to the QP is that

$$U(K) = -H_3^{-1}H_2$$

which can be written as:

$$u(k|k) = - \begin{bmatrix} 1 & 0 & \dots & 0 \end{bmatrix} (H^T W_1 H + W_2)^{-1} H^T W_1 G \mathbf{x}(k)$$
  
=  $-K \mathbf{x}(k)$ 

which is just a state feedback controller.

- Can apply this gain to the system and check the eigenvalues.

 What can we say about the stability of MPC when the constraints are active? <sup>33</sup>

- Depends a lot on the terminal cost and the terminal constraints. $^{34}$ 

- Classic result:<sup>35</sup> Consider a MPC algorithm for a linear system with constraints. Assume that there are terminal constraints:
  - $-\mathbf{x}(k+N|k) = 0$  for predicted state  $\mathbf{x}$
  - $-\mathbf{u}(k+N|k)=0$  for computed future control u

Then if the optimization problem is feasible at time k,  $\mathbf{x} = 0$  is stable.

**Proof:** Can use the performance index J as a Lyapunov function.

- Assume there exists a feasible solution at time k and cost  $J_k$
- Can use that solution to develop a feasible candidate at time k+1, by simply adding  $\mathbf{u}(k+N+1) = 0$  and  $\mathbf{x}(k+N+1) = 0$ .
- Key point: can estimate the candidate controller performance

$$\begin{split} \tilde{J}_{k+1} &= J_k - \{ \| \mathbf{z}(k|k) \|_{R_{\text{zz}}} + \| \mathbf{u}(k|k) \|_{R_{\text{uu}}} \} \\ &\leq J_k - \{ \| \mathbf{z}(k|k) \|_{R_{\text{zz}}} \} \end{split}$$

- This candidate is suboptimal for the MPC algorithm, hence J decreases even faster  $J_{k+1} \leq \tilde{J}_{k+1}$
- Which says that J decreases if the state cost is non-zero (observability assumptions)  $\Rightarrow$  but J is lower bounded by zero.
- Mayne *et al.* [2000] provides excellent review of other strategies for proving stability – different terminal cost and constraint sets

<sup>&</sup>lt;sup>33</sup> "Tutorial: model predictive control technology," Rawlings, J.B. American Control Conference, 1999. pp. 662-676

<sup>&</sup>lt;sup>34</sup>Mayne, D.Q., J.B. Rawlings, C.V. Rao and P.O.M. Scokaert, "Constrained Model Predictive Control: Stability and Optimality," Automatica, 36, 789-814 (2000).

<sup>&</sup>lt;sup>35</sup>A. Bemporad, L. Chisci, E. Mosca: "On the stabilizing property of SIORHC", Automatica, vol. 30, n. 12, pp. 2013-2015, 1994.

• Consider a system similar to the Quansar helicopter $^{36}$ 



Figure by MIT OpenCourseWare.

- There are 2 control inputs voltage to each fan  $V_f$ ,  $V_b$
- A simple dynamics model is that:

$$\begin{aligned} \ddot{\theta}_e &= K_1 (V_f + V_b) - T_g / J_e \\ \ddot{\theta}_r &= -K_2 \sin(\theta_p) \\ \ddot{\theta}_p &= K_3 (V_f - V_b) \end{aligned}$$

and there are physical limits on the elevation and pitch:

$$-0.5 \le \theta_e \le 0.6 \qquad -1 \le \theta_p \le 1$$

• Model can be linearized and then discretized  $T_s = 0.2$ sec.



 $<sup>^{36}</sup>$ ISSN 02805316 ISRN LUTFD2/TFRT-<br/> -7613- -SE MPC<br/>tools 1.0  $\,$  Reference Manual Johan Akesson Department of Automatic Control Lund Institute of Technology January 2006





Figure 16.6: Response with  ${\cal N}=25$